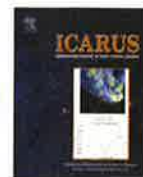




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# Mutual orbits and masses of six transneptunian binaries

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## ABSTRACT

We present Hubble Space Telescope observations of six binary transneptunian systems: 2000 QL<sub>251</sub>, 2003 TJ<sub>58</sub>, 2001 XR<sub>254</sub>, 1999 OJ<sub>4</sub>, (134860) 2000 OJ<sub>67</sub>, and 2004 PB<sub>108</sub>. The mutual orbits of these systems are found to have periods ranging from 22 to 137 days, semimajor axes ranging from 2360 to 10500 km, and eccentricities ranging from 0.09 to 0.55. These orbital parameters enable estimation of system masses ranging from 0.2 to  $9.7 \times 10^{18}$  kg. For reasonable assumptions of bulk density (0.5 to 2.0 g cm<sup>-3</sup>), the masses can be combined with visible photometry to constrain sizes and albedos. The resulting albedos are consistent with an emerging picture of the dynamically “Cold” Classical sub-population having relatively high albedos, compared with comparably-sized objects on more dynamically excited orbits.

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## 1. Introduction

As with binaries in other astrophysical settings, transneptunian binaries (TNBs) offer a way of measuring fundamental, but otherwise unobtainable physical properties such as masses and densities (e.g., Noll et al., 2008a). TNBs also offer potential insights into their dynamical environments over past history. They are relatively fragile systems which can be disrupted or have their mutual orbits altered by various external influences (e.g., Petit and Mousis, 2004; Kern and Elliot, 2006). To capitalize on the opportunities offered by TNBs will require knowledge of a large sample of their mutual orbits, enabling statistical comparisons among various sub-groupings. Toward that end, this paper describes Hubble Space Telescope (HST) observations of six TNB systems as listed in Table 1, leading to determination of the periods, semimajor axes, and eccentricities of their mutual orbits.

## 2. Data acquisition, reduction, and orbit determination

Data used in this paper were acquired by HST programs 9386, 10514, 10800, and 11178, extending over Cycles 11 through 16. Various instruments, filters, and observing strategies were employed by these programs, but crucially, they all took multiple sequential images, dithered to improve sampling of the point spread function (PSF) of the telescope, to allow identification and exclusion of

Table 1

Heliocentric orbital characteristics of the six TNB systems.

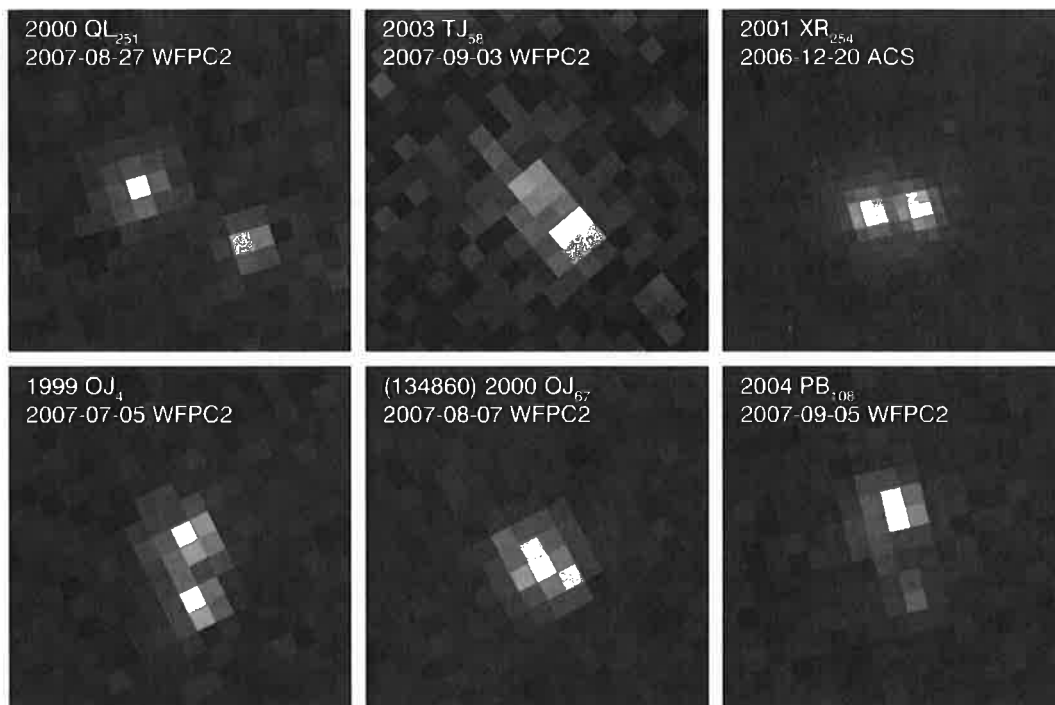
TNB system number & designation	Mean heliocentric orbital elements <sup>a</sup>			Dynamical class <sup>b</sup>
	$a_{\odot}$ (AU)	$e_{\odot}$	$i_{\odot}$ (°)	
2000 QL <sub>251</sub>	47.8	0.208	5.83	Resonant 2:1
2003 TJ <sub>58</sub>	44.5	0.094	1.31	Classical
2001 XR <sub>254</sub>	43.0	0.024	2.66	Classical
1999 OJ <sub>4</sub>	38.1	0.018	2.61	Classical
(134860) 2000 OJ <sub>67</sub>	42.9	0.014	1.33	Classical
2004 PB <sub>108</sub>	45.1	0.107	19.19	Scattered

<sup>a</sup> Averaged over a 10 Myr integration, with  $i_{\odot}$  relative to the invariable plane, as described by Elliot et al. (2005).

<sup>b</sup> Classifications are according to the Deep Ecliptic Survey system (DES; Elliot et al., 2005). The Gladman et al. (2008) system would identify 2004 PB<sub>108</sub> as a Classical object.

bad pixels and cosmic ray strikes, and to enable direct calculation of the precision of the photometry and relative astrometry derived from the data. Example single-frame images are shown in Fig. 1.

Two instruments were involved in the discovery of the binary nature and initial epoch astrometry of the TNBs in this paper. The Near Infrared Camera and Multi-Object Spectrometer (NICMOS; see Barker et al., 2007) was used by Noll et al. Cycle 11 program 9386



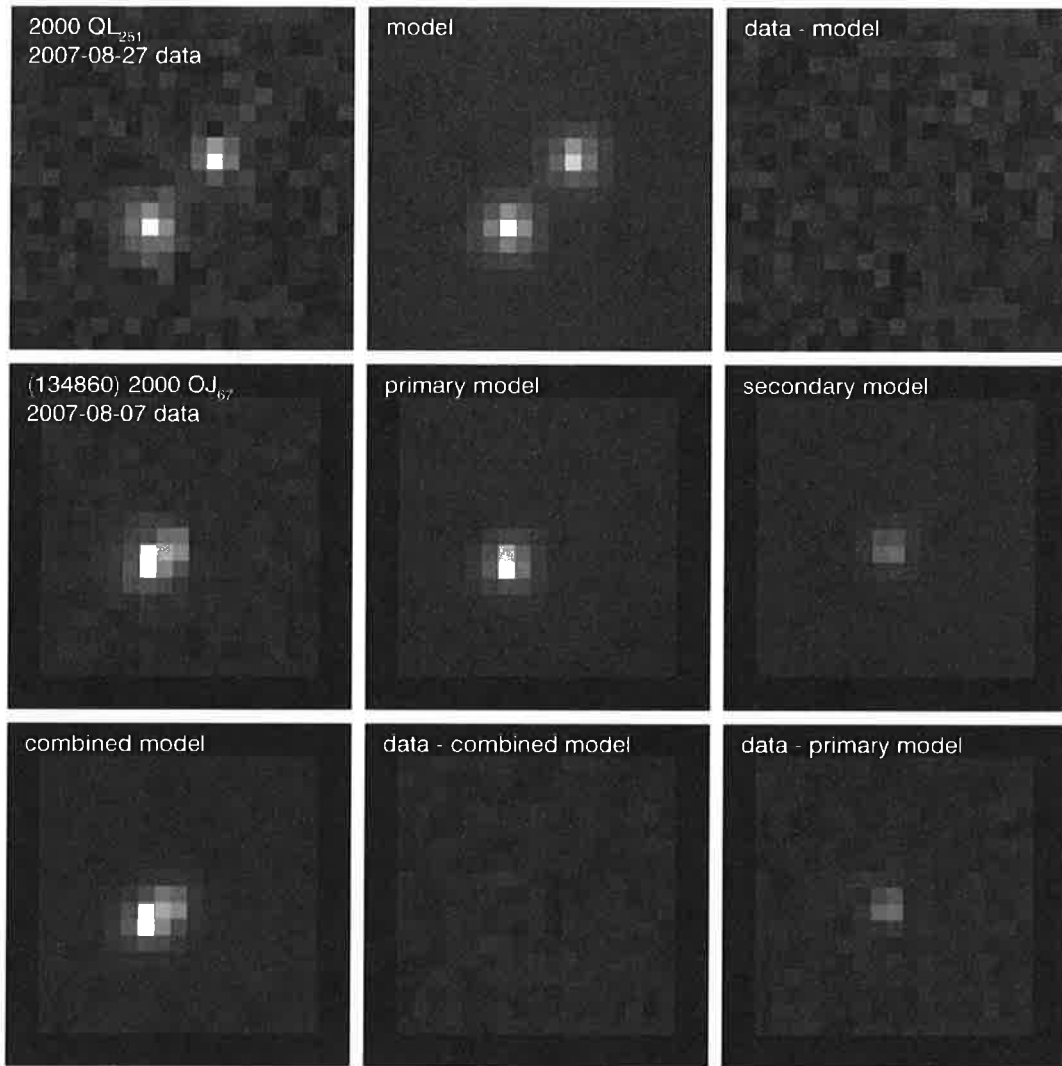
**Fig. 1.** Examples of single-frame HST images of our six TNB targets, stretched linearly and projected to sky-plane geometry with North up and East to the left. For scale, the width of this entire montage is 2 arcsec. Dithering is crucial with WFPC2 to fully sample the PSF, so for each visit we record images using four separate pointings per filter. Note the smaller, more distorted shapes of the ACS/HRC pixels when projected to the sky plane (upper right sub-panel). These smaller pixels do a better job of sampling the PSF.

That program serendipitously discovered a number of binary companions, providing single epoch relative astrometry at a relatively coarse pixel scale (Stephens and Noll, 2006). The Advanced Camera for Surveys (ACS; Ford et al., 1996) was used in a pair of searches for TNO binary companions by Noll et al. in Cycles 14 and 15 (programs 10514 and 10800; see Noll et al., 2008b for details). To maximize sensitivity to faint and close-in companions, those programs used the high-throughput *CLEAR* filter combination of ACS's High Resolution Camera (HRC). The capabilities of ACS/HRC for relative astrometry of point sources separated by a small fraction of an arcsec benefit from the unusually fine pixel scale and an especially thorough calibration effort (e.g., Sirianni et al., 2005; Pavlovsky et al., 2006; Boffi et al., 2007).

Follow-up observations were obtained by Grundy et al. Cycle 16 program 11178. The untimely failure of ACS forced this program to use the older Planetary Camera of the Wide Field and Planetary Camera 2 (WFPC2/PC; McMaster et al., 2008). WFPC2/PC has no clear filter, so most observations were done using *F606W*, the most sensitive available filter (additional observations using the *F814W* filter to obtain colors are the subject of a companion paper, Benecchi et al., 2009, which describes in detail the extraction of photometry from WFPC2 images). Observing overheads with WFPC2/PC are considerably higher than with ACS, the pixel scale is coarser and does not sample the telescope's PSF as well as ACS/HRC does, and charge transfer efficiencies are inferior. These factors combine to reduce the signal precision and number of frames which can be obtained in a single HST orbit and to reduce the precision of the relative astrometry which can be extracted from them.

Our data reduction procedures rely on the detailed knowledge of HST's exceptionally stable PSF expressed in the Tiny Tim software package (e.g., Krist and Hook, 2004). For each image frame, we simultaneously fit a pair of PSFs representing the primary and secondary body. The scatter of the measurements from the multiple dithered frames obtained during a single visit is used to estimate uncertainties on the mean for that visit. Details of our processing pipelines for NICMOS and for ACS/HRC data are published elsewhere (Stephens and Noll, 2006; Grundy et al., 2008; Benecchi et al., 2009). For WFPC2/PC data, the procedure is comparable. Fitting is done in the data frame, minimizing  $\chi^2$  for a  $21 \times 21$  pixel region extracted from the full  $800 \times 800$  data frame (see Fig. 2; smaller boxes are occasionally used for tighter pairs to avoid background clutter). The downhill simplex "amoeba" algorithm (Nelder and Mead, 1965; Press et al., 1992) is used to simultaneously adjust the parameters of a synthetic image to minimize  $\chi^2$  with respect to the data image. The synthetic image is produced from a high resolution Tiny Tim PSF generated for a solar spectral distribution and the appropriate region of the WFPC2/PC array. This PSF is re-scaled and re-sampled to simulate point sources at appropriate fluxes and locations. Real images often appear slightly blurred in comparison with synthetic PSFs. A variety of factors may contribute to this image degradation, including telescope focus variations ("breathing"), redder than Solar object colors, finite sizes of the observed objects, tracking errors, and perhaps other effects not yet recognized. To approximately account for these effects without adding numerous additional free parameters, we convolve the synthetic image with a 2-dimensional, rotationally-symmetric Gaussian smearing kernel, with a full width at half maximum generally smaller than a pixel. Note that this smearing Gaussian is distinct from the pixel-response function which we also include in generating the synthetic image. The sky background is fit separately, and for higher flux objects, the width

and 1600 nm, respectively. Detailed filter response information can be found at [http://www.stsci.edu/hst/HST\\_overview/instruments](http://www.stsci.edu/hst/HST_overview/instruments).



**Fig. 2.** Example single frame PSF fits for two of the WFPC2 images shown in Fig. 1. From left to right across the top row are the  $21 \times 21$  pixel region of the data frame used in the fitting, the two-PSF model, and the residuals (data minus model, with average background added back in), all scaled the same, linearly. In the bottom two rows we used smaller  $17 \times 17$  pixel sub-images because the separation between the two bodies was smaller. This example shows the primary and secondary components of the model image separately, and combined, along with residuals after subtracting the combined model and after subtracting only the primary component, leaving an image of the fainter secondary. Goodness of fit for these models are indicated by  $\chi^2_v = 0.8$  for the model in the top row and  $\chi^2_v = 0.9$  for the model in the bottom two rows.

of the smearing Gaussian is also fit separately, iterating between fitting the background level, the width of the smearing Gaussian, and the PSF locations and amplitudes until nothing changes. Fitting these components separately was found to improve numerical stability. These procedures usually result in goodness of fit to the image around unity, as measured by  $\chi^2_v$  ( $\chi^2$  reduced by the number of degrees of freedom). However, for other TNBs much brighter than the six subjects of this paper,  $\chi^2_v$  tends to be greater than one. Final fitted pixel positions are converted to relative astrometry by means of the ORIENTAT header keyword and the F606W image scale of  $45.5543 \pm 0.0050$  mas pixel $^{-1}$  (McMaster et al., 2008; for F814W, the value is  $45.5743 \pm 0.0050$  mas pixel $^{-1}$ ). We identify the brighter object as the “primary” and the fainter one as the “secondary.” This identification occasionally needs to be reversed after we learn more about the system, but that is a simple matter of a sign change. Uncertainties in the visit mean astrometry are estimated from the scatter of the measurements, imposing a 1 mas

floor to avoid over-weighting visits which could happen to have small measurement scatters by chance.

For each TNB system discussed in this paper, from four to six follow-up visits were needed to determine the orbital period. Mean astrometric and photometric measurements and estimated  $1-\sigma$  uncertainties are listed in Table 2.

To minimize the number of follow-up observations required, we used a scheduling strategy inspired by the Monte Carlo Statistical Ranging approach of Virtanen et al. (2008, and references therein). Details are in a companion paper (Grundy et al., 2008). Briefly, random orbits consistent with existing observations were collected and used to explore what regions of orbital element space were consistent with the data. These Monte Carlo orbit collections were also used to determine what future observation times would be best for excluding the remaining possible orbital solutions, thereby homing in on the true one. Finally, the orbit collections were used to evaluate whether or not we had arrived at a unique solution, or still needed additional observations. For each TNB in this pa-

Table 2

Observational circumstances, relative astrometry, and photometry.

Object and mean UT observation date <sup>a</sup>	HST instrument	$r^b$ (AU)	$\Delta^b$ (AU)	$g^b$ (°)	$\Delta x^c$ (arcsec)	$\Delta y^c$ (arcsec)	$V_{\text{primary}}^d$ (mag)	$V_{\text{secondary}}^d$ (mag)	
2000 QL <sub>251</sub>									
2006/07/25	10 <sup>h</sup> .7690	ACS/HRC	38.821	38.281	1.28	−0.24089(045)	−0.10082(050)	–	–
2007/07/15	22 <sup>h</sup> .7517	WFPC2/PC	38.943	38.576	1.40	−0.10209(286)	−0.16995(124)	23.950(22)	24.243(51)
2007/07/19	9 <sup>h</sup> .8100	WFPC2/PC	38.944	38.523	1.37	−0.05006(118)	−0.15815(219)	23.885(10)	24.077(14)
2007/08/05	11 <sup>h</sup> .7683	WFPC2/PC	38.950	38.287	1.14	−0.02229(264)	+0.08140(189)	24.030(05)	23.823(32)
2007/08/27	17 <sup>h</sup> .9405	WFPC2/PC	38.958	38.064	0.70	−0.23335(194)	−0.12592(176)	23.708(47)	23.961(48)
2008/08/25	8 <sup>h</sup> .6061	WFPC2/PC	39.088	38.221	0.77	+0.07422(133)	−0.06801(148)	23.785(21)	23.913(20)
2003 TJ <sub>58</sub>									
2006/11/22	5 <sup>h</sup> .8297	ACS/HRC	40.913	39.966	0.40	+0.10010(122)	+0.06001(092)	–	–
2007/08/12	1 <sup>h</sup> .6364	WFPC2/PC	40.939	41.427	1.24	+0.09071(305)	−0.00587(239)	24.335(39)	24.780(57)
2007/09/03	2 <sup>h</sup> .4072	WFPC2/PC	40.942	41.077	1.40	+0.08640(182)	+0.10062(147)	24.327(46)	24.844(47)
2007/10/22	16 <sup>h</sup> .8825	WFPC2/PC	40.947	40.295	1.06	−0.05749(143)	+0.16160(227)	24.205(25)	24.722(49)
2007/11/20	6 <sup>h</sup> .2408	WFPC2/PC	40.950	40.023	0.48	−0.10744(215)	+0.07768(254)	24.125(20)	24.663(49)
2001 XR <sub>254</sub>									
2006/12/20	6 <sup>h</sup> .6769	ACS/HRC	44.184	43.254	0.42	−0.10693(098)	+0.01285(052)	–	–
2007/09/17	11 <sup>h</sup> .1853	WFPC2/PC	44.174	44.588	1.18	−0.30433(135)	+0.15170(107)	23.305(52)	23.522(52)
2007/09/18	2 <sup>h</sup> .4936	WFPC2/PC	44.174	44.578	1.19	−0.30475(194)	+0.15598(100)	23.114(12)	23.763(40)
2007/09/21	7 <sup>h</sup> .6645	WFPC2/PC	44.174	44.527	1.21	−0.31466(155)	+0.16542(100)	23.282(52)	23.720(35)
2007/10/09	2 <sup>h</sup> .6269	WFPC2/PC	44.173	44.229	1.29	−0.30501(140)	+0.19840(156)	23.292(67)	23.600(52)
2007/12/05	9 <sup>h</sup> .9554	WFPC2/PC	44.171	43.371	0.75	+0.10316(100)	+0.01377(188)	23.440(63)	23.378(44)
2007/12/28	14 <sup>h</sup> .6145	WFPC2/PC	44.170	43.208	0.27	−0.0178(180)	−0.02445(703)	22.892(68)	23.70(13)
1999 OJ <sub>4</sub>									

**Table 3**Orbital parameters and 1- $\sigma$  uncertainties for 2000 QL<sub>251</sub>.

Parameter		Orbit 1 ( $\chi^2 = 0.37$ )	Orbit 2 ( $\chi^2 = 1.9$ )
<b>Fitted elements<sup>a</sup></b>			
Period (days)	$P$	56.459 $\pm$ 0.018	56.443 $\pm$ 0.017
Semimajor axis (km)	$a$	4991 $\pm$ 17	5014 $\pm$ 16
Eccentricity	$e$	0.4874 $\pm$ 0.0062	0.4867 $\pm$ 0.0060
Inclination <sup>b</sup> (deg)	$i$	127.78 $\pm$ 0.62	45.62 $\pm$ 0.66
Mean longitude <sup>b</sup> at epoch <sup>c</sup> (deg)	$\epsilon$	146.1 $\pm$ 1.5	110.70 $\pm$ 0.74
Longitude of asc. node <sup>b</sup> (deg)	$\Omega$	109.5 $\pm$ 1.1	71.2 $\pm$ 1.1
Longitude of periapsis <sup>b</sup> (deg)	$\tilde{\omega}$	151.70 $\pm$ 0.97	116.9 $\pm$ 1.5
<b>Derived parameters</b>			
Standard gravitational parameter $GM_{\text{sys}}$ (km <sup>3</sup> day <sup>-2</sup> )	$\mu$	0.2063 $\pm$ 0.0021	0.2092 $\pm$ 0.0020
System mass (10 <sup>18</sup> kg)	$M_{\text{sys}}$	3.090 $\pm$ 0.031	3.135 $\pm$ 0.030
Orbit pole right ascension <sup>b</sup> (deg)	$\alpha_{\text{pole}}$	19.5 $\pm$ 1.0	341.2 $\pm$ 1.1
Orbit pole declination <sup>b</sup> (deg)	$\delta_{\text{pole}}$	-37.78 $\pm$ 0.61	+44.38 $\pm$ 0.67
Orbit pole ecliptic longitude (deg)	$\lambda_{\text{pole}}$	359.9 $\pm$ 1.1	5.7 $\pm$ 1.2
Orbit pole ecliptic latitude (deg)	$\beta_{\text{pole}}$	-41.83 $\pm$ 0.58	+47.16 $\pm$ 0.57
Next mutual event season		2083	2084

<sup>a</sup> Elements are for secondary relative to primary. The average sky plane residual for Orbit 1 is 0.36 mas and the maximum is 0.90 mas.

<sup>b</sup> Referenced to J2000 equatorial frame.

<sup>c</sup> The epoch is Julian date 2454200.0 (2007 April 9 12:00 UT).

**Table 4**Orbital parameters and 1- $\sigma$  uncertainties for 2003 TJ<sub>58</sub>.

Parameter		Orbit 1 ( $\chi^2 = 1.9$ )	Orbit 2 ( $\chi^2 = 6.4$ )
<b>Fitted elements<sup>a</sup></b>			
Period (days)	$P$	137.32 $\pm$ 0.19	137.32 $\pm$ 0.19
Semimajor axis (km)	$a$	3799 $\pm$ 54	3728 $\pm$ 44
Eccentricity	$e$	0.528 $\pm$ 0.011	0.529 $\pm$ 0.011
Inclination <sup>b</sup> (deg)	$i$	38.1 $\pm$ 2.1	96.1 $\pm$ 2.0
Mean longitude <sup>b</sup> at epoch <sup>c</sup> (deg)	$\epsilon$	56.0 $\pm$ 2.2	31.9 $\pm$ 1.9
Longitude of asc. node <sup>b</sup> (deg)	$\Omega$	194.6 $\pm$ 4.2	150.8 $\pm$ 2.8
Longitude of periapsis <sup>b</sup> (deg)	$\tilde{\omega}$	84.56 $\pm$ 0.73	61.9 $\pm$ 3.6
<b>Derived parameters</b>			
Standard gravitational parameter $GM_{\text{sys}}$ (km <sup>3</sup> day <sup>-2</sup> )	$\mu$	0.01537 $\pm$ 0.00066	0.01453 $\pm$ 0.00052
System mass (10 <sup>17</sup> kg)	$M_{\text{sys}}$	2.304 $\pm$ 0.099	2.178 $\pm$ 0.078
Orbit pole right ascension <sup>b</sup> (deg)	$\alpha_{\text{pole}}$	104.6 $\pm$ 4.3	60.8 $\pm$ 2.9
Orbit pole declination <sup>b</sup> (deg)	$\delta_{\text{pole}}$	+51.9 $\pm$ 2.2	-6.1 $\pm$ 2.1
Orbit pole ecliptic longitude (deg)	$\lambda_{\text{pole}}$	100.2 $\pm$ 3.1	57.2 $\pm$ 3.0
Orbit pole ecliptic latitude (deg)	$\beta_{\text{pole}}$	+29.0 $\pm$ 2.1	-26.3 $\pm$ 2.0
Next mutual event season		2098	2059

<sup>a</sup> Elements are for secondary relative to primary. The average sky plane residual for Orbit 1 is 1.1 mas and the maximum is 2.2 mas.

**Table 7**  
Orbital parameters and 1- $\sigma$  uncertainties for (134860) 2000 OJ<sub>67</sub>.

Parameter		Orbit 1 ( $\chi^2 = 2.3$ )	Orbit 2 ( $\chi^2 = 2.5$ )
<b>Fitted elements<sup>a</sup></b>			
Period (days)	$P$	$22.0412 \pm 0.0040$	$22.0412 \pm 0.0036$
Semimajor axis (km)	$a$	$2361 \pm 36$	$2352 \pm 35$
Eccentricity	$e$	$0.090 \pm 0.022$	$0.085 \pm 0.021$
Inclination <sup>b</sup> (deg)	$i$	$84.6 \pm 3.0$	$73.8 \pm 2.9$
Mean longitude <sup>b</sup> at epoch <sup>c</sup> (deg)	$\epsilon$	$71.0 \pm 3.2$	$22.4 \pm 3.0$
Longitude of asc. node <sup>b</sup> (deg)	$\Omega$	$272.9 \pm 3.1$	$212.2 \pm 3.3$
Longitude of periastris <sup>b</sup> (deg)	$\omega$	$39 \pm 14$	$349 \pm 17$
<b>Derived parameters:</b>			
Standard gravitational parameter $GM_{\text{sys}}$ ( $\text{km}^3 \text{day}^{-2}$ )	$\mu$	$0.1433 \pm 0.0066$	$0.1417 \pm 0.0063$
System mass ( $10^{18} \text{ kg}$ )	$M_{\text{sys}}$	$2.146 \pm 0.099$	$2.123 \pm 0.094$
Orbit pole right ascension <sup>b</sup> (deg)	$\alpha_{\text{pole}}$	$182.9 \pm 3.1$	$122.2 \pm 3.2$
Orbit pole declination <sup>b</sup> (deg)	$\delta_{\text{pole}}$	$+5.4 \pm 3.1$	$+16.2 \pm 3.0$
Orbit pole ecliptic longitude (deg)	$\lambda_{\text{pole}}$	$180.5 \pm 3.0$	$120.9 \pm 2.9$
Orbit pole ecliptic latitude (deg)	$\beta_{\text{pole}}$	$+6.1 \pm 3.2$	$-3.8 \pm 3.2$
Next mutual event season		2104	2056

<sup>a</sup> Elements are for secondary relative to primary. The average sky plane residual for Orbit 1 is 1.3 mas and the maximum is 1.9 mas.

<sup>b</sup> Referenced to J2000 equatorial frame.

<sup>c</sup> The epoch is Julian date 2454000.0 (2006 September 21 12:00 UT).

**Table 8**  
Orbital parameters and 1- $\sigma$  uncertainties for 2004 PB<sub>108</sub>.

Parameter		Orbit 1 ( $\chi^2 = 2.2$ )	Orbit 2 ( $\chi^2 = 6.6$ )
<b>Fitted elements<sup>a</sup></b>			
Period (days)	$P$	$97.017 \pm 0.070$	$97.077 \pm 0.069$
Semimajor axis (km)	$a$	$10400 \pm 130$	$10550 \pm 130$
Eccentricity	$e$	$0.4372 \pm 0.0074$	$0.4455 \pm 0.0074$
Inclination <sup>b</sup> (deg)	$i$	$89.0 \pm 1.1$	$106.55 \pm 0.99$
Mean longitude <sup>b</sup> at epoch <sup>c</sup> (deg)	$\epsilon$	$168.49 \pm 0.63$	$59.2 \pm 1.5$
Longitude of asc. node <sup>b</sup> (deg)	$\Omega$	$121.99 \pm 0.75$	$30.19 \pm 0.86$
Longitude of periastris <sup>b</sup> (deg)	$\omega$	$351.92 \pm 0.53$	$242.1 \pm 1.6$
<b>Derived parameters:</b>			
Standard gravitational parameter $GM_{\text{sys}}$ ( $\text{km}^3 \text{day}^{-2}$ )	$\mu$	$0.632 \pm 0.023$	$0.660 \pm 0.025$
System mass ( $10^{18} \text{ kg}$ )	$M_{\text{sys}}$	$9.47 \pm 0.35$	$9.88 \pm 0.37$
Orbit pole right ascension <sup>b</sup> (deg)	$\alpha_{\text{pole}}$	$31.99 \pm 0.75$	$300.19 \pm 0.85$
Orbit pole declination <sup>b</sup> (deg)	$\delta_{\text{pole}}$	$+1.0 \pm 1.1$	$-16.55 \pm 0.98$
Orbit pole ecliptic longitude (deg)	$\lambda_{\text{pole}}$	$30.16 \pm 0.89$	$298.90 \pm 0.90$
Orbit pole ecliptic latitude (deg)	$\beta_{\text{pole}}$	$-11.2 \pm 1.0$	$+3.91 \pm 0.91$
Next mutual event season		2103	2039

<sup>a</sup> Elements are for secondary relative to primary. The average sky plane residual for Orbit 1 is 1.6 mas and the maximum is 3.4 mas.

<sup>b</sup> Referenced to J2000 equatorial frame.

<sup>c</sup> The epoch is Julian date 2454200.0 (

**Table 9**

Adopted orbital elements and photometric properties.

TNB system	Period $P$ (days) <sup>a</sup>	Semimajor axis $a$ (km) <sup>a</sup>	Eccentricity $e$ <sup>a</sup>	Combined $H_V$ <sup>b</sup> (mag)	$\Delta_{\text{mag}}$ <sup>b</sup> (mag)
2000 QL <sub>251</sub>	56.451 ± 0.025	5002 ± 27	0.4871 ± 0.0065	7.148 ± 0.016	0.073 ± 0.048
2003 TJ <sub>58</sub>	137.32 ± 0.19	3768 ± 85	0.529 ± 0.011	7.448 ± 0.024	0.519 ± 0.031
2001 XR <sub>254</sub>	125.61 ± 0.13	9271 ± 130	0.5516 ± 0.0091	6.030 ± 0.017	0.415 ± 0.064
1999 OJ <sub>4</sub>	84.115 ± 0.038	3267 ± 60	0.365 ± 0.012	7.344 ± 0.012	0.088 ± 0.020
(134860) 2000 OJ <sub>67</sub>	22.0412 ± 0.0040	2357 ± 40	0.088 ± 0.025	6.471 ± 0.010	0.530 ± 0.073
2004 PB <sub>108</sub>	97.05 ± 0.10	10480 ± 200	0.441 ± 0.012	7.083 ± 0.016	1.321 ± 0.032

<sup>a</sup> Orbital element values for the orbit of the secondary about the primary along with uncertainties here are chosen such that symmetric error bars encompass the 1- $\sigma$  uncertainties of the two mirror orbit solutions for each TNB.

<sup>b</sup> Average absolute  $V$  magnitude  $H_V$  of the combined light from both bodies and average  $V$  magnitude difference between primary and secondary  $\Delta_{\text{mag}}$  computed from the ensemble of WFFC2  $F606W$  observations as described by a companion paper (Benecchi et al., 2009) assuming  $G = 0.15$  in the  $H$  and  $G$  magnitude system (Bowell et al., 1989). These numbers differ somewhat from numbers in Benecchi et al. (2009, Table 3) because those numbers were based only on the sub-set of visits consisting of interleaved observations in  $F606W$  and  $F814W$  filters (typically just one visit per system in Cycle 16 program 11178).

**Table 10**

Derived quantities.

Object designation	System mass <sup>a</sup> , $M_{\text{sys}}$ (kg)	Primary radius (km) <sup>b</sup>	$V$ geometric albedo, $A_p$ <sup>b</sup>	Orbital angular momentum, $J_{\text{orb}}/J'$	Hill radius, $r_H$ (km)	$a/r_H$
2000 QL <sub>251</sub>	$(3.112 \pm 0.051) \times 10^{18}$	57–91	0.04–0.10	1.7–2.1	460,000	0.011
2003 TJ <sub>58</sub>	$(2.25 \pm 0.15) \times 10^{17}$	25–40	0.16–0.41	2.2–2.7	200,000	0.019
2001 XR <sub>254</sub>	$(4.00 \pm 0.17) \times 10^{18}$	65–104	0.09–0.22	2.1–2.7	550,000	0.017
1999 OJ <sub>4</sub>	$(3.91 \pm 0.22) \times 10^{17}$	29–46	0.12–0.31	1.9–2.4	230,000	0.014
2000 OJ <sub>67</sub>	$(2.14 \pm 0.11) \times 10^{18}$	53–85	0.09–0.23	1.2–1.5	450,000	0.005
2004 PB <sub>108</sub>	$(9.68 \pm 0.57) \times 10^{18}$	93–148	0.02–0.05	1.6–2.0	710,000	0.015

<sup>a</sup> System masses  $M_{\text{sys}}$  are based on the CODATA 2006 value of the gravitational constant  $G = 6.6742 \times 10^{-11} \text{ m}^3 \text{ s}^{-2} \text{ kg}^{-1}$ .

<sup>b</sup> Radii and albedos are based on  $M_{\text{sys}}$  and a plausible density range of 0.5 to 2.0  $\text{g cm}^{-3}$ , assuming both bodies share the same albedo and density. Higher densities would permit smaller radii and higher albedos, while lower densities would allow larger radii and lower albedos.

function; Press et al., 1992), that orbit can be formally excluded. At present, none of the mirror solutions for the six TNBs in this paper reach that threshold. The closest is 1999 OJ<sub>4</sub>, for which Orbit 2 can be excluded at 2.3- $\sigma$  confidence. In the mean time, for each TNB system we adopt  $P$ ,  $a$ , and  $e$  values intermediate between the two mirror solutions, with uncertainties inflated to encompass the full range of 1- $\sigma$  uncertainties for the two solutions. These adopted values appear in Table 9, along with mean  $V$ -band photometric properties.

### 3. Results

System masses for our TNBs are computed according to

$$M_{\text{sys}} = \frac{4\pi^2 a^3}{G P^2}, \quad (1)$$

where  $G$  is the gravitational constant (we use the CODATA 2006 value  $G = 6.6742 \times 10^{-11} \text{$







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